

ASTR 120
Fall 2006
Homework Assignment No. 2 – Solutions

1. (a) The Moon's sidereal angular velocity (rate of rotation):

$$\omega = \frac{360^\circ}{27.3 \text{ days}} = \frac{360^\circ}{27.3 \text{ days} \times 1440 \text{ min/day}} = 9.16 \times 10^{-3} \text{ min}^{-1}$$

$$t = \frac{.5^\circ}{\omega} = \frac{.5^\circ}{9.16 \times 10^{-3} \text{ min}^{-1}} = 54.6 \text{ minutes} = .91 \text{ hrs}$$

- (b)

$$\theta = \omega t = 9.16 \times 10^{-3} \text{ min}^{-1} \times (12 \text{ hrs} \times 60 \text{ min/hr}) = 6.59^\circ$$

Yes. As per the hinted fig. (1-10) such a change in angle is easily visible/measurable with the naked eye.

2. A sidereal month is ~ 27.3 days and a synodic ~ 29.5 .

So there are $(365.25 \text{ days/year})/27.3 \text{ days} \sim 13.38$ sidereal months in a year

and there are $(365.25 \text{ days/year})/29.5 \text{ days} = 12.38$ synodic months in a year.

Thus there is ~ 1 more sidereal than synodic month per year.

3. $F = m a = 3 \text{ kg} \times 2 \text{ m/s}^2 = 6 \text{ Newtons}$

$$\text{and } \frac{2F}{m} = 2a = 4 \text{ m/s}^2$$

4. (a) Total major axis = 2 AU + 6 AU, so semi-major axis, $a = \frac{2+6}{2} = 4 \text{ AU}$

(b) Kepler's 2nd Law: $P^2 = a^3 = (4 \text{ AU})^3 = 64$

$$\text{So } P = \sqrt{64} = 8 \text{ yrs}$$

Extra "A":

(a) Ceres' semi-major axis: $a = \frac{2.99+2.55}{2} = 2.77 \text{ AU}$

(b) $P^2 = a^3 = (2.77 \text{ AU})^3 = 21.25$ So $P = \sqrt{21.25} = 4.61 \text{ yrs}$

5. (a) Again, use Kepler's 2nd Law: $P = 64 \text{ yrs}$

$$\text{So... } a = P^{\frac{2}{3}} = 64^{\frac{2}{3}} = 16 \text{ AU}$$

(b) Aphelion distance (d_{Ap}) + Perihelion distance (d_{Per}) = 2 a

$$\text{In this case: } d_{Per} = 2a - d_{Ap} = (2 \times 16) - 31.5 = .5 \text{ AU}$$

Extra "B": $a = P^{\frac{2}{3}} = 2500^{\frac{2}{3}} = 184.2 \text{ AU}$

$$d_{Ap} = 2a - d_{Per} = (2 \times 184.2) - .9 = 367.5 \text{ AU}$$

6. Gravitational acceleration: $g = \frac{GM}{R^2}$

On this other planet $M_{PlanetX} = 4M_{\oplus}$ and $R_{PlanetX} = 4R_{\oplus}$ so

$$g_{PlanetX} = \frac{GM_{PlanetX}}{R_{PlanetX}^2} = \frac{G(4M_{\oplus})}{(4R_{\oplus})^2} = \frac{g_{\oplus}}{4}$$

Extra “C”:

Phobos:

$$M_{Phobos} = 1.1 \times 10^{16} \text{ kg}, D_{Phobos} = 23 \text{ km (I chose the middle value...)}$$

Convert these to “Earth Masses” and “Earth Radii”:

$$M_{Phobos} = \frac{1.1 \times 10^{16} \text{ kg}}{5.9 \times 10^{24} \text{ kg}/M_{\oplus}} = 1.86 \times 10^{-9} M_{\oplus}$$

$$R_{Phobos} = .5 \times D_{Phobos} = \frac{(.5 \times 23 \text{ km})}{6378.137 \text{ km}/R_{\oplus}} = 1.8 \times 10^{-3} R_{\oplus}$$

$$g_{Phobos} = \frac{GM_{Phobos}}{R_{Phobos}^2} = \frac{G(1.86 \times 10^{-9} M_{\oplus})}{(1.8 \times 10^{-3} R_{\oplus})^2} = 5.74 \times 10^{-4} g_{\oplus}$$

Callisto:

$$M_{Callisto} = 1.077 \times 10^{23} \text{ kg} = \frac{1.077 \times 10^{23} \text{ kg}}{5.9 \times 10^{24} \text{ kg}/M_{\oplus}} = 1.83 \times 10^{-2} M_{\oplus}$$

$$D_{Callisto} = 4800 \text{ km}; R_{Callisto} = .5 \times D_{Callisto} = \frac{(.5 \times 4800 \text{ km})}{(6378.137 \text{ km}/R_{\oplus})} = 3.76 \times 10^{-1} R_{\oplus}$$

$$g_{Callisto} = \frac{GM_{Callisto}}{R_{Callisto}^2} = \frac{G(1.83 \times 10^{-2} M_{\oplus})}{(3.76 \times 10^{-1} R_{\oplus})^2} = 1.29 \times 10^{-1} g_{\oplus}$$

Titan:

$$M_{Titan} = 1.34 \times 10^{23} \text{ kg} = \frac{1.34 \times 10^{23} \text{ kg}}{5.9 \times 10^{24} \text{ kg}/M_{\oplus}} = 2.27 \times 10^{-2} M_{\oplus}$$

$$D_{Titan} = 5150 \text{ km}; R_{Titan} = .5 \times D_{Titan} = \frac{(.5 \times 5150 \text{ km})}{(6378.137 \text{ km}/R_{\oplus})} = 4.04 \times 10^{-1} R_{\oplus}$$

$$g_{Titan} = \frac{GM_{Titan}}{R_{Titan}^2} = \frac{G(2.27 \times 10^{-2} M_{\oplus})}{(8.075 \times 10^{-1} R_{\oplus})^2} = 0.139 g_{\oplus}$$

7.

$$F_{Saturn} = \frac{GM_{\odot} M_{Saturn}}{d_{Saturn}^2} = \frac{GM_{\odot} (100M_{\oplus})}{(10d_{\oplus})^2} = F_{\oplus}$$

$$a = \frac{F}{m} \quad \text{so} \quad a_{Saturn} = \frac{F_{Saturn}}{m_{Saturn}} \quad \text{and} \quad a_{\oplus} = \frac{F_{\oplus}}{m_{\oplus}}$$

$$\text{Since } M_{Saturn} = 100M_{\oplus} \quad a_{Saturn} = \frac{a_{\oplus}}{100}$$

Extra Credit:

1. The period is one day.

2. Use Newton’s form of Kepler’s 2nd Law: $P^2 = \frac{4\pi^2}{GM_{\oplus}} a^3$ (we can ignore $M_{Satellite}$ in the denominator since $M_{\oplus} \gg M_{Satellite}$) so

$$a = \left(P^2 \frac{GM_{\oplus}}{4\pi^2} \right)^{1/3} = \left((1 \text{ day} \times 8.64 \times 10^4 \text{ sec/day})^2 \frac{(6.67 \times 10^{-8} \text{ dyn cm}^2 \text{ g}^{-2})(5.9 \times 10^{27} \text{ g})}{4\pi^2} \right)^{1/3} = 4.2 \times 10^9 \text{ cm} = 4.2 \times 10^4 \text{ km}$$

3. If its orbit was inclined with respect to Earth's equator then the point directly under the satellite would not remain the same; it would oscillate in latitude even if the orbit was synchronous with the Earth's rotation.

Think about the Sun's motion in the sky; it changes in latitude (i.e. the sub-solar point does) due to the fact that Earth's axis is tilted 23.5 degrees.

Also, don't be confused into thinking that only a geosynchronous orbit can have a one day period; that's not correct.